

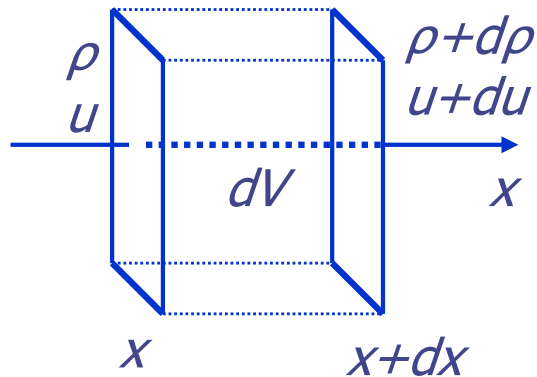
# Spherical accretion

Literature:

Frank, King, Raine, *Accretion Power in  
Astrophysics*, 2002

# Conservation laws

- ◆ Gasdynamics can be applied if particles collide many times before crossing the region, i.e. the mean free path  $\lambda \ll$  typical size  $L$ . Gas can be described by  $P, \rho, T, u$ .
- ◆ Mass conservation. Continuity equation:



$$\frac{\partial(\rho dx)}{\partial t} = \frac{\partial}{\partial t}[\rho u - (\rho + d\rho)(u + du)] = \frac{\partial}{\partial t}[-ud\rho - \rho du]$$

$$\frac{\partial \rho}{\partial t} + u \frac{\partial \rho}{\partial x} + \rho \frac{\partial u}{\partial x} = 0 \Rightarrow \frac{\partial \rho}{\partial t} + \frac{\partial(\rho u)}{\partial x} = 0$$

if  $\frac{\partial}{\partial t} = 0 \Rightarrow \rho u = \text{const}$

- ◆ In 3D:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \vec{u}) = 0$$

# Momentum conservation. Euler's equation.

- We assume that the gas pressure is the only force acting on the gas.
- In astrophysics other forces are often important: magnetic, gravitational, viscous, and radiation pressure forces

$$\frac{\partial(\rho u dx)}{\partial t} = [\rho u^2 - (\rho + d\rho)(u + du)^2] + P - (P + dP)$$

$$\frac{\partial(\rho u)}{\partial t} = -\frac{\partial P}{\partial x} - u^2 \frac{\partial \rho}{\partial x} - 2\rho u \frac{\partial u}{\partial x}$$

$$\rho \frac{\partial u}{\partial t} + u \frac{\partial \rho}{\partial t} = \rho \frac{\partial u}{\partial t} + u \left[ -u \frac{\partial \rho}{\partial x} - \rho \frac{\partial u}{\partial x} \right] \quad \Leftarrow \text{from mass conservation}$$

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} = -\frac{1}{\rho} \frac{\partial P}{\partial x}$$

For  $\frac{\partial}{\partial t} = 0$  and using mass conservation  $\rho u = \text{const} \Rightarrow \rho u^2 + P = \text{const}$

- In 3D:

$$\frac{\partial \vec{u}}{\partial t} + (\vec{u} \cdot \nabla) \vec{u} = -\frac{1}{\rho} \nabla P + \frac{\vec{F}}{m}$$

- ◆ **Conservation of energy.** Most complicated equation. Consider two simple limiting cases.
- ◆ **Adiabatic flow.** No radiation losses or input, no conduction. Volume element perform work on surrounding. The equation of state:

$$P = K\rho^\gamma, \quad \gamma = c_p / c_v \text{ is the ratio of specific heats.}$$
$$\gamma = 5/3 \text{ monoatomic gas, } = 7/5 \text{ for diatomic gas.}$$

- ◆ **Isothermal flow.** Temperature is determined by heating=cooling balance.  $T$  is adjusted much faster than dynamical time-scale.

$$T = \text{const}$$

# Sound waves

◆ Instead of the energy equation:  $P = K\rho^\gamma$   $\gamma=5/3$  - adiabatic,  $\gamma=1$  - isothermal

◆ Consider gas at rest and small perturbations

$$P = P_0 = \text{const}, \rho = \rho_0 = \text{const}, u = 0$$

$$P = P_0 + P_1, \rho = \rho_0 + \rho_1, u = u_1$$

$$P_1 = K\gamma\rho_0^{\gamma-1}\rho_1 = \gamma\frac{P_0}{\rho_0}\rho_1, \text{ define } c_s^2 \equiv \gamma\frac{P_0}{\rho_0}$$

◆ Continuity equation:

$$\frac{\partial(\rho_0 + \rho_1)}{\partial t} + u_1 \frac{\partial\rho_0}{\partial x} + (\rho_0 + \rho_1) \frac{\partial u_1}{\partial x} = 0 \Rightarrow \frac{\partial\rho_1}{\partial t} + \rho_0 \frac{\partial u_1}{\partial x} = 0 \Rightarrow \frac{1}{\rho_0} \frac{\partial^2\rho_1}{\partial t^2} + \frac{\partial^2 u_1}{\partial t \partial x} = 0$$

◆ Momentum equation:

$$\frac{\partial u_1}{\partial t} + u_1 \frac{\partial u_1}{\partial x} = -\frac{1}{\rho} \frac{\partial(P_0 + P_1)}{\partial x} \Rightarrow \frac{\partial u_1}{\partial t} + \frac{1}{\rho_0} \frac{\partial P_1}{\partial x} = \frac{\partial u_1}{\partial t} + \frac{1}{\rho_0} c_s^2 \frac{\partial\rho_1}{\partial x} = 0$$

$$\Rightarrow \frac{\partial^2 u_1}{\partial t \partial x} + \frac{c_s^2}{\rho_0} \frac{\partial^2 \rho_1}{\partial x^2} = 0$$

◆ The wave equation

$$\frac{\partial^2 \rho_1}{\partial t^2} - c_s^2 \frac{\partial^2 \rho_1}{\partial x^2} = 0, \quad \frac{\partial^2 u_1}{\partial t^2} - c_s^2 \frac{\partial^2 u_1}{\partial x^2} = 0$$

◆ Solution:  $\rho_1 = f(x \pm c_s t)$  if  $u_0 \neq 0$  sound propagates with  $u_0 \pm c_s$ .

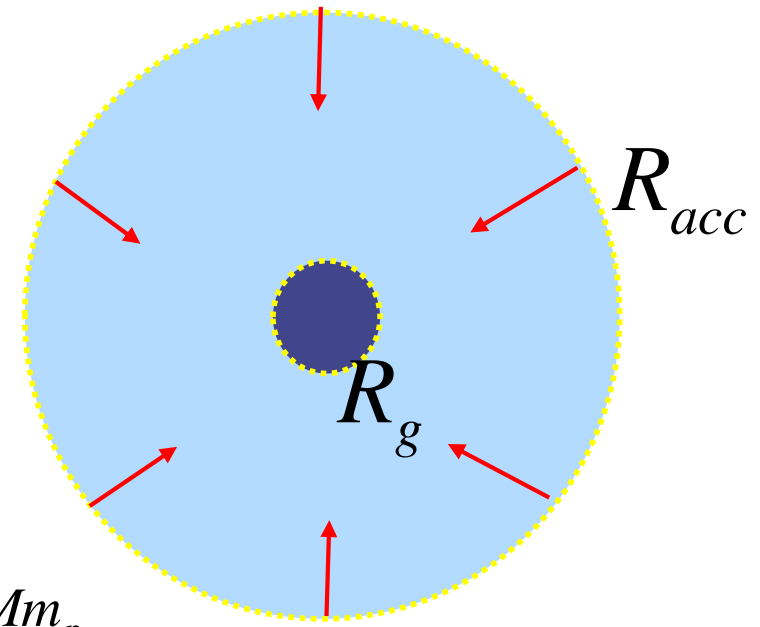
◆ Note

$$c_s \propto \rho^{(\gamma-1)/2}, \quad \gamma=5/3, \quad c_s \propto \rho^{1/3} \text{ adiabatic;}$$

$$\gamma=1, \quad c_s = \text{const} - \text{isothermal}$$

# Spherical accretion

- ◆ Given a large cloud with  $T(\infty)$  and  $\rho(\infty)$ . Black hole (BH) in the cloud is an ideal “vacuum cleaner”.
- ◆  $R_g = 2GM/c^2$  is the radius of the absorbing surface, the event horizon.
- ◆  $R_{acc}$  is the radius of which the BH gravitational pull dominates over thermal motions in the cloud:
- ◆ For  $M = M_\odot$  and  $c_s = 10$  km/s,  $R_{acc} = 1.3 \cdot 10^{14}$  cm = 10 au.



$$\frac{GMm_p}{R_{acc}} \approx kT \Rightarrow$$

$$R_{acc} \approx \frac{GMm_p}{kT} = \frac{GM}{kT / m_p} \approx \frac{GM}{P / \rho} \approx \frac{GM}{c_s^2}$$

$$\left[ P = \frac{\rho}{\mu m_p} kT \Rightarrow c_s^2 = \frac{\partial P}{\partial \rho} \right]$$

- What is the accretion rate  $\dot{M}$ ?
- What is the inflow velocity profile  $V(r)$ ?
- What is the gas temperature profile  $T(r)$ ?
- What is the emission from the flow?

- ◆ Consider steady-state solution for a spherically symmetric flow.
- ◆ Simple equations of hydrodynamics

Continuity equation

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \vec{V}) = 0 \Rightarrow \frac{1}{r^2} \frac{d}{dr} (r^2 \rho V) = 0$$

$$(1) \frac{d\dot{M}(r)}{dr} = 0, \quad \dot{M} = 4\pi r^2 \rho(r) V(r)$$

Euler equation

$$\rho \left( \frac{\partial \vec{V}}{\partial t} + (\vec{V} \cdot \nabla) \vec{V} \right) = -\nabla P + \vec{f} \Rightarrow$$

$$\rho \frac{dV}{dt} = -\frac{dP}{dr} + f$$

External force  
(gravitational) per cm<sup>3</sup>

- ◆  $V$ - speed of a given element. But if we are interested in velocity profile at a given space coordinate, we have to make a transformation:

$$dV = \frac{\partial V}{\partial t} dt + \frac{\partial V}{\partial r} dr$$

- ◆ Change of velocity consists of two parts.

- change of the velocity in a given point during time  $dt$
- Difference in velocities (at the same moment  $t$ ) in two points separated by  $dr$ , which is the distance the elements moves in  $dt$ :

$$\frac{dV}{dt} = \frac{\partial V}{\partial t} + \frac{dr}{dt} \frac{\partial V}{\partial r}$$

In steady-state

$$\rho \left[ \frac{\partial V}{\partial t} + V \frac{\partial V}{\partial r} \right] = -\frac{dP}{dr} + f \Rightarrow \rho V \frac{dV}{dr} = -\frac{dP}{dr} - \frac{GM\rho}{r^2} \quad (2)$$

## Equation of state

$$(3) \quad P = \frac{\rho k T}{\mu m_H} \quad - \text{ ideal gas, } \mu \text{ is the mean mass per particle}$$

$\mu = 1$  for neutral H,  $\mu = 1/2$  for ionized H.

## The Mach number

$$\mathbf{M} = \frac{V}{c_s}, \quad \text{where } c_s \text{ is the sound speed}$$

$\mathbf{M} < 1$ - subsonic flow. In the limit  $\mathbf{M} \ll 1$  hydrostatics

$\mathbf{M} > 1$ - supersonic flow. In the limit  $\mathbf{M} \gg 1$  free-fall

## Polytropic flow

$$P = K \rho^\gamma$$

Examples : adiabatic flow,  $\gamma = 5/3$  ; isothermal flow ( $T = \text{const}$ ),  $\gamma = 1$ .

$$c_s^2 \equiv \frac{\partial P}{\partial \rho} = \gamma \frac{P}{\rho}, \quad c_s \text{ - sound speed}$$



$$\left. \begin{aligned} \frac{dP}{dr} &= \frac{dP}{d\rho} \frac{d\rho}{dr} = c_s^2 \frac{d\rho}{dr} \\ (1) \Rightarrow \frac{d\rho}{dr} &= -\frac{\rho}{V} \frac{dV}{dr} - \frac{2\rho}{r} \\ (2) \Rightarrow \rho V \frac{dV}{dr} &= -\frac{dP}{dr} - \frac{GM\rho}{r^2} \end{aligned} \right\} \Rightarrow V \frac{dV}{dr} [1 - \mathbf{M}^{-2}] = \frac{2c_s^2}{r} \left[ 1 - \frac{GM}{2rc_s^2} \right] \quad (4)$$

$$\text{at } r \rightarrow \infty, \mathbf{M} \rightarrow 0 \quad [V \rightarrow 0, c_s \rightarrow c_s(\infty)]$$

$$\text{at } r \rightarrow R_g, \mathbf{M} > 1 \quad \left[ V \frac{dV}{dr} < 0, 1 - \frac{c^2}{c_s^2} < 0 \Rightarrow 1 - \mathbf{M}^{-2} > 0 \right]$$

Hence at some point  $\mathbf{M} = 1$  - transonic flow

Let  $\mathbf{M} = 1$  at  $r = r_s$  (sonic point)

Then from (4) at  $r = r_s$   $\frac{dV}{dr}(r_s) = \infty$  or  $c_s^2(r_s) = \frac{GM}{2r_s}$  ← We choose a physical solution

[The general "regularity" condition for any transonic flow. After expressing explicitly  $V \frac{dV}{dr} = \frac{N(r)}{D(r)}$ . There exists such  $r_s$  that  $N(r) = D(r) = 0$ ]

For a polytropic gas, one can integrate the Euler equation (2) to get the Bernoulli equation

$$\frac{V^2}{2} + \frac{c_s^2}{\gamma-1} - \frac{GM}{r} = \text{const}$$

The constant can be found by considering the limit  $r \rightarrow \infty$ . Then  $V \rightarrow 0, GM/r \rightarrow 0$  and  $\text{const} = c_s^2(\infty)/(\gamma-1)$ .

We thus get a set of algebraic equations describing the accretion problem:

$$(1) \quad \dot{M} = 4\pi r^2 \rho V = \text{const} \quad - \text{to be found (continuity)}$$

$$(2) \quad \frac{V^2}{2} + \frac{c_s^2}{\gamma-1} - \frac{GM}{r} = \frac{c_s^2(\infty)}{\gamma-1} \quad (\text{Bernoulli equation})$$

$$(3) \quad \left[ \frac{\rho}{\rho(\infty)} \right]^{\gamma-1} = \left[ \frac{c_s}{c_s(\infty)} \right]^2 \quad (\text{the polytropic condition})$$

$$\left. \begin{aligned} (4) \quad c_s(r_s) &= V(r_s) \\ (5) \quad c_s^2(r_s) &= \frac{GM}{2r_s} \end{aligned} \right\} \quad (\text{the regularity conditions})$$

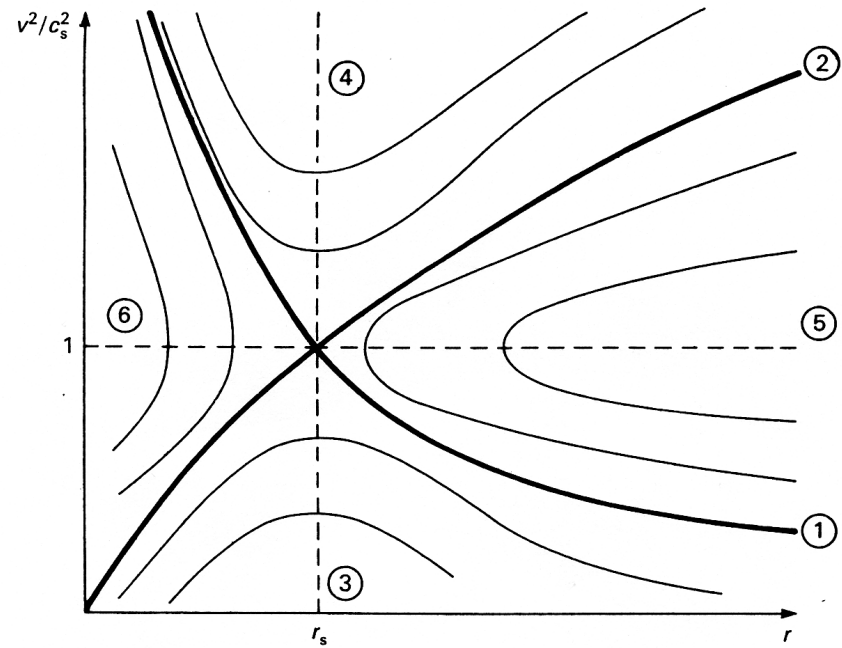


Figure 1. Spherical adiabatic gas flows  $v^2(r)/c_s^2(r)$  in the gravitational field of a star. For  $v < 0$  these are accretion flows, while for  $v > 0$  they are winds or 'breezes'. The two trans-sonic solutions 1, 2 divide the remaining solutions into the families 3–6 described in the text.

# The case $\gamma=5/3$ (adiabatic accretion)

At home you will show that

$$(a) \quad \dot{M} = \pi G^2 M^2 \frac{\rho(\infty)}{c_s^3(\infty)} \left[ \frac{2}{5-3\gamma} \right]^{(5-3\gamma)/2(\gamma-1)}$$

$$(b) \quad r_s = \frac{GM}{c_s^2(\infty)} \frac{5-3\gamma}{4}$$

When  $\gamma \rightarrow 5/3$ ,  $r_s$  decreases from  $GM / c_s^2(\infty)$  to  $r_s = 0$ .

As a result, the transonic region get "stretched out" into the whole range of radii  $r \ll GM / c_s^2(\infty)$ .

The flow is near the regularity condition.

$$\left. \begin{aligned} V(r) &\approx c_s(r) \\ c_s^2(r) &\approx \frac{GM}{2r} \end{aligned} \right] \text{ at } r \ll \frac{GM}{c_s^2(\infty)} \quad \boxed{\mathbf{M} \approx 1, V(r) \approx \sqrt{\frac{GM}{2r}}}$$

The accretion velocity is about virial,  $V_{vir} \approx \sqrt{GM/r}$ , and the gas temperature is about virial  $kT_{vir} \approx GMm_p/r$  (since  $c_s^2 \approx kT_{vir}/m_p$ ). The gas gets heated with decreasing  $r$  ( $T \propto 1/r$ ) owing to "adiabatic heating",  $dU = -PdV_{vol}$  (1st law of thermodynamics).

The temperature reaches  $kT \approx m_e c^2$  at

$$r_* \approx \frac{GM}{c^2} \frac{m_p}{m_e} \approx R_g \frac{m_p}{m_e}$$

Inside  $r_*$  the electrons are relativistic (their internal energy  $\gg m_e c^2$ ).

Then the adiabatic index changes.

One can show that internal energy density of a polytropic gas satisfies the relation  $\varepsilon = P / (\gamma - 1)$

Ionized non-relativistic plasma,  $kT \ll m_e c^2$

$$\left. \begin{aligned} \varepsilon &= \frac{3}{2} nkT + \frac{3}{2} nkT = 3nkT \\ P &= nkT + nkT = 2nkT \end{aligned} \right\} \Rightarrow \gamma = 5/3.$$

Ionized relativistic plasma,  $m_e c^2 \ll kT \ll m_p c^2$

$$\left. \begin{aligned} \varepsilon &= \frac{3}{2} nkT + 3nkT = \frac{9}{2} nkT \\ P &= nkT + \frac{1}{3} 3nkT = 2nkT \end{aligned} \right\} \Rightarrow \gamma = 13/9.$$

At  $r_*$  the index  $\gamma$  changes from  $5/3$  to  $13/9$ . Since  $13/9 < 5/3$ ,  $r_s$  shifts from 0 to a value  $\leq r_*$  and the flow becomes supersonic at  $r \ll r_*$ .

# Emission from the accretion flow

Most of the energy is released in the very vicinity of the BH, at radii  $r \approx R_g$

The innermost region makes the main contribution to the observed luminosity.

Therefore, to estimate the luminosity and spectrum, one needs to know the flow parameters at  $r \approx R_g \ll r_*$ . Here the gas is nearly free-falling ( $\mathbf{M} \gg 1$ ) with velocity  $\approx c$ .

The density can be found from the continuity equation (mass conservation):

$$\dot{M} = 4\pi r^2 \rho V \Rightarrow \rho = \frac{\dot{M}}{4\pi r^2 V} \approx \frac{\dot{M}}{4\pi R_g^2 c}$$

The temperature is determined from the Bondi solution with  $\gamma=13/9$  which should match the solution with  $\gamma=5/3$  at  $r \approx r_*$ .

At home you prove:  $\frac{T(r)}{T_*} \approx \left(\frac{r_*}{r}\right)^{2/3}$ , at  $r \ll r_*$ .

Thus at  $R_g$ ,  $T(R_g) \approx m_e c^2 \left(\frac{m_p}{m_e}\right)^{2/3} \approx 70 \text{ MeV}$ .

The observed spectrum is produced by a plasma cloud of size  $R_g$  with temperature  $T(R_g)$  and density  $\rho(R_g)$ .

For the densities and temperatures of interest, the dominant emission is free-free emission (if no magnetic field).

$$\Lambda_{ff} = \Lambda_{ei} + \Lambda_{ee} \text{ erg/cm}^3 / \text{s (emission rate per unit volume).}$$

$$\Lambda_{ei} = 12\alpha r_e^2 n^2 ckT \left[ \frac{3}{2} + \ln \frac{2kT}{m_e c^2} - 0.577 \right]$$

$$\Lambda_{ee} = 24\alpha r_e^2 n^2 ckT \left[ \frac{5}{4} + \ln \frac{2kT}{m_e c^2} - 0.577 \right], \text{ for } kT \gg m_e c^2.$$

Here  $\alpha = \frac{e^2}{\hbar c} = 1/137$  - the fine structure constant,  $r_e$  is the classical electron radius.

The luminosity from the accretion flow:

$$L_{ff} = \int_{R_g}^{\infty} \Lambda_{ff} 4\pi r^2 dr \approx \Lambda_{ff} \frac{4\pi}{3} R_g^3 \approx 10^{21} \left( \frac{n_{\infty}}{1 \text{ cm}^{-3}} \right)^2 \left( \frac{T_{\infty}}{10^4 \text{ K}} \right)^{-3} \left( \frac{M}{M_{sun}} \right)^3 \text{ erg/s}$$

The resulting (free-free) spectrum is  $L_{\nu} \propto \nu^0$  with an exponential cutoff at  $h\nu \approx kT \approx 70 \text{ MeV}$ .

Note: the luminosity is low  $\frac{L_{ff}}{L_{Edd}} \approx 10^{-17}$ .

In the presence of magnetic field, an important emission mechanism is synchrotron emission. The synchrotron emissivity of hot relativistic ( $kT \gg m_e c^2$ ) electrons:

$$\Lambda_{syn} = \frac{16}{3} \frac{e^2}{c} \left( \frac{eB}{m_e c} \right)^2 \left( \frac{kT}{m_e c^2} \right)^2 n = \frac{16}{3} r_e^2 c B^2 n \left( \frac{kT}{m_e c^2} \right)^2 \text{ erg/cm}^3/\text{s}$$

The plausible magnetic field (Shvartsman 1971), is the "equipartition" field:

$$\frac{B^2}{8\pi} \approx \frac{GM\rho}{r}.$$

Show that the resulting synchrotron luminosity from the accretion flow:

$$L_{syn} \approx 10^{27} \left( \frac{n_\infty}{1 \text{ cm}^{-3}} \right) \left( \frac{T_\infty}{10^4 \text{ K}} \right)^{-3} \left( \frac{M}{M_{sun}} \right)^3 \text{ erg/s}$$

The luminosity from spherical accretion is thus low even in the presence of the equipartition magnetic field.

The radiative efficiency of accretion

$$\eta = \frac{L_{syn}}{\dot{M}c^2} \leq 10^{-4}.$$

- ◆ Where this theory is applicable? In situations where a black hole is surrounded by low angular momentum gas
  - Center of the Milky Way : the central black hole is surrounded by gas originating from stellar winds. Accretes with little angular momentum onto black hole.
  - Center of elliptical galaxies : interstellar medium is hot ( $\sim 5$  million K; X-ray emitting) and accretes with low angular momentum onto central supermassive black hole.
  
- ◆ In both these cases the black holes are very quiescent, i.e., they produce little electromagnetic radiation.